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Barcelona Supercomputing Center Centro Nacional de Supercomputación

Computatinal Structural Mechanics using High Performance Computing

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Amade Day Winter 2023

Barcelona Supercomputing Center-Centro Nacional de Supercomputación (BSC-CNS)

BSC-CNS OBJECTIVES



Supercomputing services to Spanish and EU researchers



R&D in Computer, Life, Earth and Engineering Sciences



PhD programme, technology transfer, public engagement

BSC-CNS is a consortium that includes

Spanish Government	60%	E AND
Catalan Government	30%	Generalitat de Catalunya Departament d'Empresa i Coneixement
Univ. Politècnica de Catalunya (UPC)	10%	UNIVERSITAT POLITECNICA DE CATALUNYA BARCELONATECH

Barcelona Supercomputing Center-Centro Nacional de Supercomputación (BSC-CNS)

- Spanish National Supercomputing Center
- Located at the Technical University of Catalonia -BarcelonaTech
- Around **800 people** from several disciplines
- Four main departments: Engineering (CASE),
 Computer Science, Life Science and Earth Science
- BSC holds MareNostrum IV (13.4 PetaFLOPS)



MareNostrum 5: Pre-exascale supercomputing

- >310 PetaFLOPS peak performance (> 310 x 10¹⁵)
- > 2PetaBytes of storage
- 400PetaBytes of active archive
- Aggregated Investment of €223 million
- + Experimental Platform to create supercomputing technology "made in Europe"







The acquisition and operation of the EuroHPC supercomputer is funded jointly by the EuroHPC Joint Undertaking, through the European Union's Connecting Europe Facility and the Horizon 2020 research and innovation programme, as well as the Participating States Spain, Portugal, and Turkey



Introduction of Alya Multiphysics code

- FEM-based Computational Mechanics code (https://alya.gitlab.bsc.es)
- Language: Fortran 2008
- Parallization strategies:
 - MPI (130K CPU-cores) + OpenMP
 - Accelerators: OpenACC-CUDA
 - Co-execution on heterogeneous systems
- Fully parallel workflow:
 - Dynamic load balance (DLB)
 - Adaptative mesh refinement (AMR)
 - Parallel multiphysics/multi-code coupling
- Alya is regularly run on PRACE Tier 0 systems and part of the European Applications Benchmark Suite (EUABSC)





Alya Towards Multiphysics







Accelerating the performance

Speed-up & Complexity Implementation

Algorithm Optimisation

Coarse-grained activities

- Node Cluster
- •



Superscalar core Accelerators

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Taskification / coloring
Synchronous to asynchronous communication
Mesh partitioning, advanced solvers
Quantifying uncertainty on simulation
Algorithm recovery from core/thread failures

Fine-grained activities

Task-based Dynamic Load Balancing
(DLB)

- Single Instruction Multiple Data (SIMD)
- Use of accelerator directives (GPU)

Example

Compression After Impact (CAI) of a MONOSTRINGER structure





Advances in solid mechanics and application examples

A parallel contact algorithm for HPC systems

Contents lists available at ScienceDirect

Computers and Structures

A parallel algorithm for unilateral contact problems

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Computers & Structures

Check for updates

Non-linear contact problem

(Hertz-Signorini-Moreau law)

 $\nabla \cdot \underline{\underline{\sigma}} + \underline{f}_{\nu} = 0 \quad \text{in } \Omega$ $\underline{\underline{\sigma}} \cdot \underline{\underline{n}} = \underline{\sigma}_{0} \quad \text{on } \Gamma_{N}$ $\underline{\underline{u}} = \underline{\underline{u}}_{0} \quad \text{on } \Gamma_{D}$ $g \ge 0, \ \sigma_{n} \le 0, \ \sigma_{n}g = 0, \ \underline{\sigma}_{t} = 0 \quad \text{on } \Gamma_{C}$







SEVIE

A parallel contact algorithm

Contents lists available at ScienceDirect

Composites Part A

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Benchmark test ASTM D7136/D7136M-20 standard

Application of the partial Dirichlet–Neumann contact algorithm to simulate low-velocity impact events on composite structures

ELSEVIER

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Multi-impact on a 5m curved panel from a passengers aircraft

Work in collaboration with Imperial College of London (ICL)

Sensor

 (\mathbf{X})

Impact from inside (skin/stringer's foot 23~31J)

Impact from fuselage-skin side 20~25J



Active sensing on a 5m panel representative of passengers aircraft

Work in collaboration with Imperial College of London (ICL)

• Sensor





Active sensing on Bay-C

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Smart virtual Fuselage representative of passengers aircraft

Work in collaboration with Imperial College of London (ICL)

Geometry details (based on curved panel)

- •D=3.4m; L=4.466m
- •Pitch: 250mm
- •Distance between frames: 620mm
- •Height frame: 100 mm
- •7 frames (6bays), 43 stringers

SHM system

•Sensors ref. PI P-876.SP1 (Circular) •1032 sensors (4 x 6 x 43) •R=5mm, h=0.5mm

Materials

•Skin/stringers (T800S/M21):

tply = 0.194mm; tcoh=0.0mm

• Lay-up skin/stringer: [45/-45/0_2/90/0]s

- Frames (Aluminum)
- Sensors (Ceramic)

Mesh details

- •46,393,066 elements
- •155 x 10⁶ DoF
- Structured mesh of HEX08 and COH08
- ·Averaged in-plane element size: 4mm

·Averaged thickness element size: 0.194 mm







Fluid-Structure-Interaction

Fluid Structure Interaction framework

Computational Fluid Dynamics (CFD)

[Lehmkuhl et al. 2018, Owen et al. 2019 (WM)]

- Eulerian formulation
- Navier-stokes equations

(1)
$$\rho_f \frac{\partial \boldsymbol{u}_f}{\partial t} + \rho_f (\boldsymbol{u}_f \cdot \nabla) \boldsymbol{u}_f - \nabla \cdot [2\mu_f \boldsymbol{\varepsilon}(\boldsymbol{u}_f)] + \nabla p = \rho_f \boldsymbol{f},$$

(2)
$$\nabla \cdot \boldsymbol{u}_f = 0,$$

Arbitrary Lagrangian-Eulerian (ALE)

[Calderer and Masud. 2010]

• Mesh movement is solved through a Laplacian equation.

Computational Solid Mechanics (CSM) [Belytscho et al. 2007 (2n edition)]

$$ho_s rac{\partial \boldsymbol{d}_s}{\partial t} = \nabla \cdot \boldsymbol{P} + \boldsymbol{b}$$

- Total Lagrange formulation in finite strains.
- Equation of the linear momentum balance.
- Newmark-Beta integration scheme.

FSI coupling algorithm

Coupling conditions



- Partitioned approach (multi-code)
- · Gauss-Seidel algorithm
- Staggered Explicit/Implicit coupling algorithm
- One-way / two way coupling
- · Data transfer: on the boundary

Benchmark test: 2-d channel with flexible wall (Mok 2001)





0.08

0.06

0.04

0.02

0

0

Tip displacement(m)

Solid	Fluid
rho =1500 kg/m3	rho = 956 kg/m3
E = 2.3e6 N/m2	mu = 0.145 kg/(m⋅s)
v = 0.45	



Fluid-Structure Interaction

Description

- Ref: WindPACT 1.5MW horizontal wind turbine blade from NREL (Malcolm and Hansen, 2002)
- · It is a conventional 3-d bladed upwind horizontal-axis wind turbine
- It is a 33m long blade (70m rotor diameter) Mid-large scale wind turbine

Mesh discretitzation

- Flow domain: Full integration linear 4-node (Tetra), 5-node (Pyr) and 6-node (Wedge)
- Flow domain: 18,011,043 elements (21.5M DoF)
- Solid domain: Full integration linear 8-node (Hexa) and 6-node (Wedge)
- Solid domain: 4,303,554 elements (14.6M DoF)













Benchmark test: 2-d channel with flexible wall (Mok 2001)



Turbulent flow past wind turbine blade





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Thank you very much for your attention!

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